

MIRROR – A Modular and Relocatable Multi-Arm Robot Demonstrator for On-orbit Large Telescope Assembly

> 17<sup>th</sup> Symposium on Advanced Space Technologies in Robotics and Automation

> > Leiden, 18-20 October 2023











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1	2	3
Introduction	Concept of Operations	Technological Demonstrator
4	5	6
Multi-Arm Robot System	Testbed	Conclusions & Perspectives



# SINCE 1987



**INTRODUCTION** 

### **INTRODUCTION – MIRROR PROJECT**

Current Context

*Large structure in space* are an *essential* and recurring element *for space exploitation and exploration* continuously increasing in size to bring increased economic and scientific benefits (e.g. James Webb telescope).

Future Context

Next structures will be too large to be launched into orbit as a single self-deploying piece that can be contained in standard launcher fairings. These **larger structures could be divided into subassemblies** (orbital replacement unit 'ORU', modules, reflector segments etc.) that are launched disassembled and **later assembled in orbit**.

esa

ESA TRP MIRROR : Multi-arm Installation Robot for Readying ORUs and Reflectors

MISROR

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Foreseen Technologies in MIRROR

& LEONARDO

The modules of a **large space telescope** will be equipped with **standard interconnects** and **a robotic system** will **relocate** over a spacecraft for **deploying** them from their stowed location to their operational location.

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### MIRROR & MAR CONCEPTS





### **MIRROR CONCEPT**





- Multi-Arm Robot (MAR)
- The MAR features a modular approach using HOTDOCKs standard interconnects.
- The MAR is composed three modules:
  - A **torso**, featuring a 1-DOF leg, a vision subsystem and a battery.
  - Two 7-DOF robotic arms.





### **MULTI-ARM ROBOT OVERVIEW**





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7

inspection

## TECHNOLOGICAL DEMONSTRATOR



applicatio

### **DEMONSTRATOR DERIVATION FROM THE MISSION SCENARIO**





### **DEMONSTRATOR OVERVIEW**



#### Flight segment

- The **MAR** capable of grasping, releasing, and transporting modules.
- The testbed: dummy spacecraft, hexagonal telescope tiles and ORU equipped with HOTDOCKs and a weight compensation device.
- Ground segment
  - The "Monitoring and Control Station" that allows users to supervise MAR's tasks.









### **MULTI-ARM ROBOT SYSTEM**

applications

spa







### **MULTI ARM ROBOT SYSTEM – STANDARD INTERCONNECT**





Central Functional interface (power/data)

 Pogo pins and pads



Mechanical interface (alignment/coupling / load transfer)

- Form fit geometry
- locking mechanism

- **7 Active interconnects** (2 per arm and 3 on the torso)
- HOTDOCK is an androgynous standard interconnects, providing:
  - Mechanical interface with locking mechanism for structural coupling
  - Electrical interface to control and distribute electrical power between connected systems
  - Bi-directional data interface, compatible with different high rate data transmission
  - Optional thermal interface

### **MULTI ARM ROBOT SYSTEM – ROBOTIC ARMS**





- Structure (limbs)
  - Symmetrical design
  - Eight subassemblies
  - Human-like arm with asymmetric joints
  - 1.85m 40kg
- Motorization (joints)
  - Integrated into the limbs
  - 7 Revolute hollow shaft joints
  - Symmetric configuration  $R \perp R \perp R \perp R \perp R \perp R \perp R \perp R$
  - Two joint sizes (165Nm and 350Nm)
- End effectors / SIs (HOTDOCKs)
- Embedded Avionics: RCU, SCUs, power switch units, data (CAN, EtherCAT) and power (48V) buses, internal cable routing.



### **MULTI ARM ROBOT SYSTEM – TORSO**





- Mechanical Structure A hub that can equip three other modules (limbs or payloads) by means of standard interconnects. Approx. 28kg.
- Two fixed active standard interconnects Static HOTDOCKs for attaching the two "robotic manipulators".
- Leg Rotary standard interconnect Attachment point to the spacecraft or to a payload.
- Perception Sensors and Lighting Modules – Monitoring the assembly and manipulation tasks.
- Embedded Avionics RCU, SCU, PDU, Battery, power (48V) and data buses (CAN, EtherCAT).









### **TESTBED**





#### **TESTBED**



- The dummy spacecraft in the MIRROR project features a home base, a storage area, and a telescope structure, with Standard Interconnects (SIs) for robot access and payloads placement.
- The weight compensation device is a passive gantry crane mechanism with a rolling bridge for X and Y movement, supported by a cable system with counterweights for Z-axis load.
- Payloads consist of two hexagonal mirror tiles (1.2m wide, 10kg each) and one parallelepiped ORU (5kg), all equipped with SIs.
- The ground segment includes a programming and control station (MCC) running on Linux and an Electrical Ground Support Equipment (EGSE) providing power, data components, and wireless communication for remote connection.



MIRROR dummy spacecraft







Payloads and ORU



### CONCLUSION





### **CONCLUSION & PERSPECTIVES**





- Introduction of the ESA TRP MIRROR project.
- Demonstrator design and preliminary integration of the MAR concept, a multiarm robotic system for on-orbit large assembly:
  - Featuring a modular approach,
  - Using Standard Interconnects,
  - Offering a complete panel of operation regardless of the robot configuration.
- Future work will focus on the testing activities of the ground demonstrator.
- Study of very large assembly in space involving modular robotics and multi-robot cooperation
  → OSIP ESA SKYBEAM project.











Multi-arm robot 2023



Walking manipulator 2021





Robotic joint 2020

### **CONTACTS AND ACKNOWLEDGMENTS**





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**MIRROR** is a project funded by the European Space Agency

